



# The Future of Astronomy will be Built: Results from the NASA in-Space Assembled Telescope (iSAT) Study

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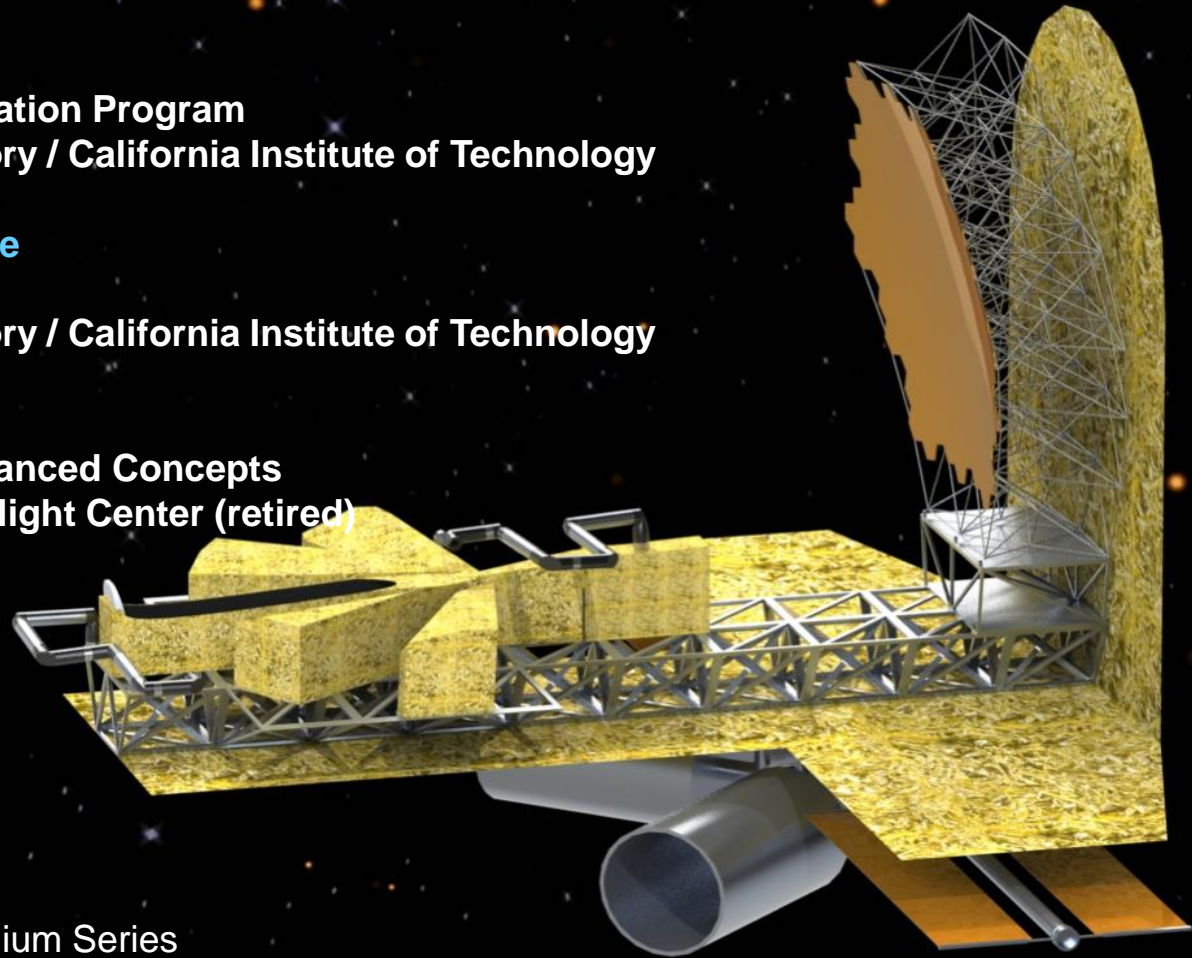
Robotics Technologist

Jet Propulsion Laboratory / California Institute of Technology

**Harley Thronson**

Senior Scientist for Advanced Concepts

NASA Goddard Space Flight Center (retired)



ExEP Technology Colloquium Series  
November 12, 2019

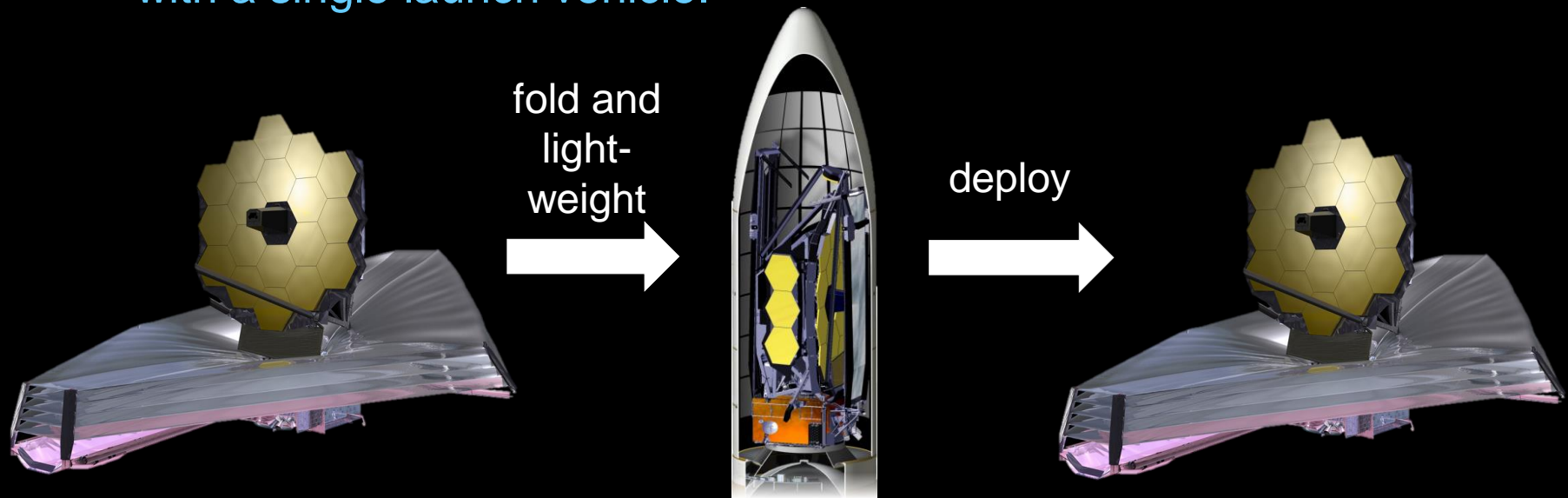
# The Future of Space Telescopes

Will benefit from...

- increased aperture size
- lowered mission risk
  - ❖ As they will be expensive as their sizes and capabilities grow
- extended mission lifetime
  - ❖ Through refueling, repairing, replenishing
  - ❖ Enabling total cost to be spread out over decades
- improved capability over time
  - ❖ Replacing instruments with more advanced ones

# The Current Approach to Space Telescopes

- Designed to meet the volume and mass requirements associated with a single launch vehicle.



- Relies on all going perfectly during deployment
  - ❖ JWST: Over 20 sequential deployment events, 40 deployable structures, 178 release mechanisms – all of which must work.
  - ❖ Repair capabilities do not yet exist other than with astronauts
  - ❖ Astronauts are not planned to go to Geo or Sun-Earth L2
- Mission lifetime is fuel or cryogen limited
  - ❖ Servicing capabilities to repair or extend lifetime not yet in place
- No new instruments - you live with what you have

One potential approach that addresses these current challenges and achieves that future is *in-Space Assembly*.



**Dr. Paul Hertz**  
**Director**  
**Astrophysics Division**  
**Science Mission Directorate**  
**NASA Headquarters**

## **Study Objective:**

- *“When is it worth assembling space telescopes in space rather than building them on the Earth and deploying them autonomously from single launch vehicles?”*

**In-Space Assembled Telescope study commenced in May 2018**

# Study Participants

<u>Name</u>	<u>Institution</u>	<u>Expertise</u>
1. Ali Azizi	NASA JPL	Metrology
2. Larry Dewell	LMC	Pointing/Stability/Control
3. Oscar Salazar	NASA JPL	Pointing/Stability/Control
4. Phil Stahl	NASA MSFC	Telescopes
5. Jon Arenberg	NGAS	Thermal/Sunshade
6. Doug McGuffey	NASA GSFC	Telescopes/SE
7. Kim Aaron	NASA JPL	Structures
8. Dave Redding	NASA JPL	Telescopes

## Study Involvement

- 72 participants
- 6 NASA Centers
- 14 private companies
- 2 gov't agencies
- 5 universities

9. Bill		
10. A		
11. B		
12. E		
13. S		
14. M		
15. M		
16. R		
17. S		
18. T		
19. P		
20. J		
21. A		
22. David Stubbs	LMC	Telescopes/Design
23. John Dorsey	NASA LaRC	Structures
24. Jeff Sokol	Ball	Mechanical/AIT
25. Atif Qureshi	SSL	Robotics SE
26. Carlton Peters	NASA GSFC	Thermal
27. Kan Yang	NASA GSFC	Thermal
28. Paul Lightsey	Ball	SE
29. Kim Mehalick	NASA GSFC	Thermal/Sunshade
30. Bo Naasz	NASA GSFC	RPO
31. Keith Havey	L3Harris	Mirror Segments
32. Harley Thronson	NASA GSFC	Mission Concepts
33. Scott Knight	Ball	Optics

<u>Name</u>	<u>Institution</u>	<u>Expertise</u>
34. John Lymer	SSL	Robotics

**Missions and Concepts**  
**JWST, HST, ISS, Restore-L, RSGS, NASA's Tipping Point, Gateway, and future large mission concepts**

45. Lynn Bowman	NASA LaRC	Programmatic
46. John Grunsfeld	ex-NASA	Astronaut

## Discipline Experts

- RPO
- telescope optics
- robotics
- structures
- sunshade
- instruments
- I&T + V&V
- launch vehicles
- orbital dynamics

72. Nick Siegler	NASA JPL	Technologist
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# Detailed Process Approach

Four-step approach

**Step 1:** Select a reference mission concept

**Step 2a:** Conduct a qualitative assessment based on experiences and lessons learned

**Step 2b:** Conduct a quantitative assessment using a grass-roots costing and risk exercise by SMEs from various subsystems

**Step 2c:** Independent parametric cost estimate for conventional telescope building

# Study Assumptions

## 1. Reference telescope:

- Non-cryogenic operating at UV/V/NIR assembled in space
- Four sizes between 5 – 20 m

## 2. Driving requirement:

- Structural stability required by coronagraphy of exo-planets

## 3. Operational destination:

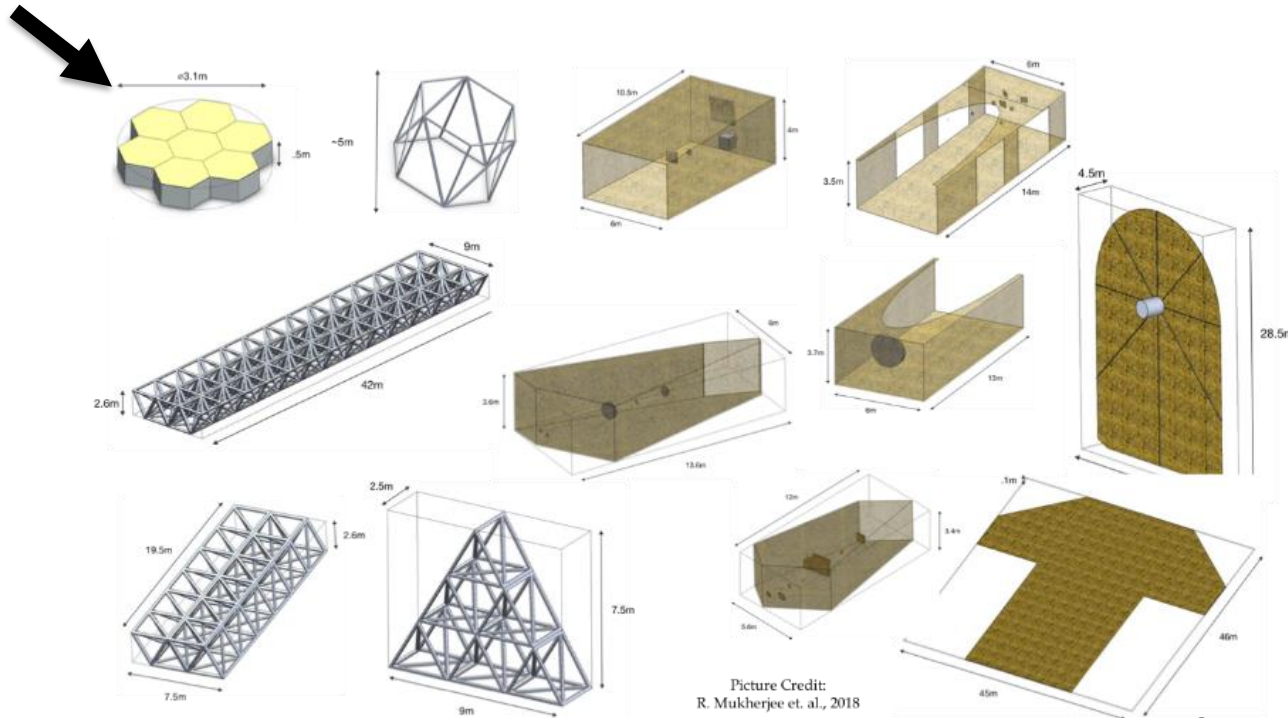
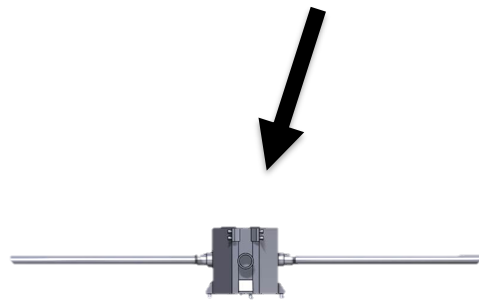
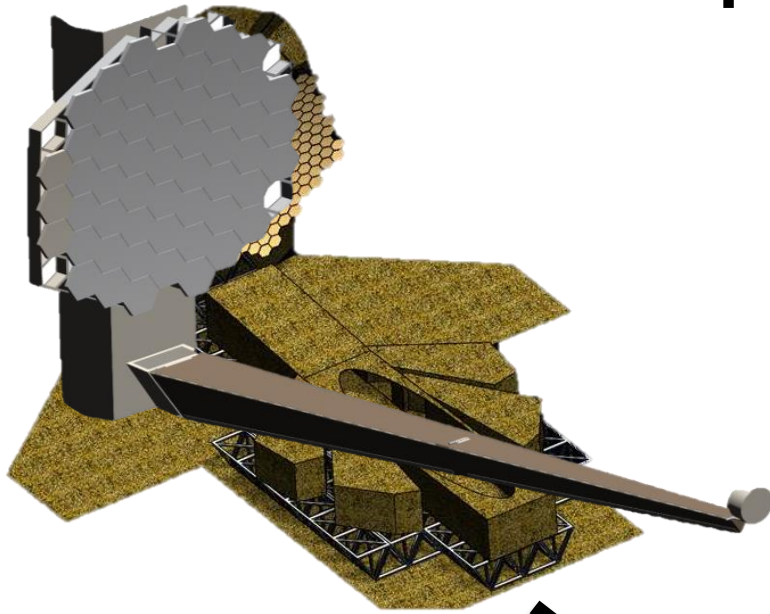
- Sun-Earth L2

## 4. Launch vehicles:

- Use of 5 m-class launch vehicle fairings



# Modularization of a Space Telescope

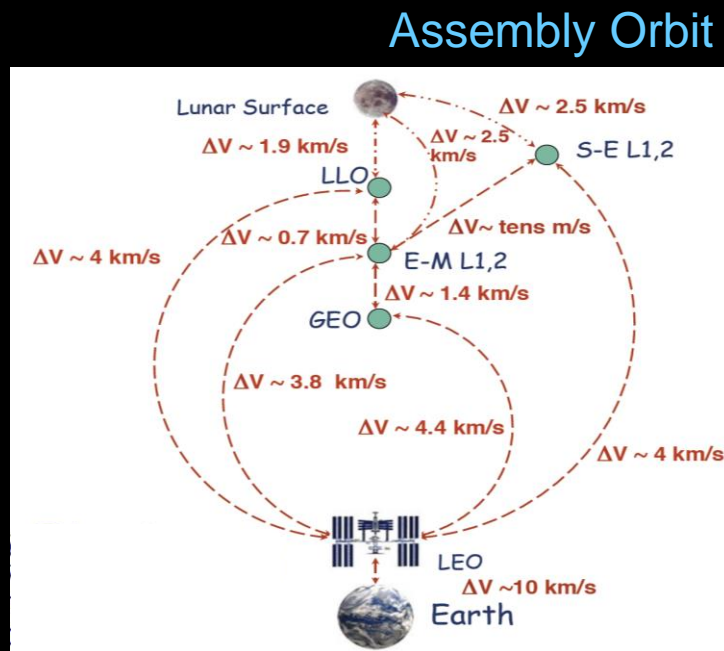
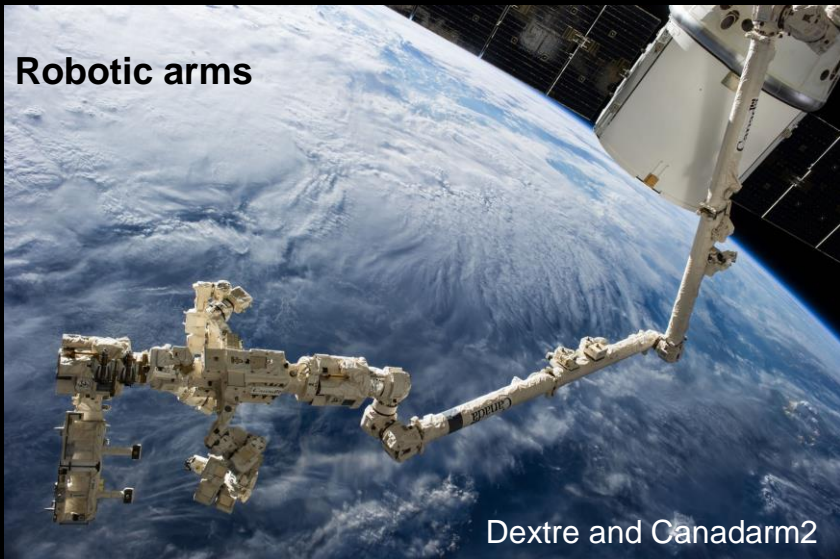


Picture Credit:  
R. Mukherjee et. al., 2018

# Reference Mission Concept

Very large option space

Assembly Agent



## Launch Vehicles

ULA's Delta IV Heavy

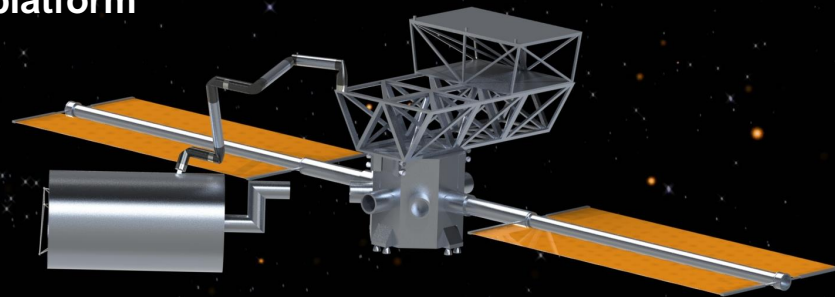
ULA's Atlas V

SpaceX's Falcon Heavy



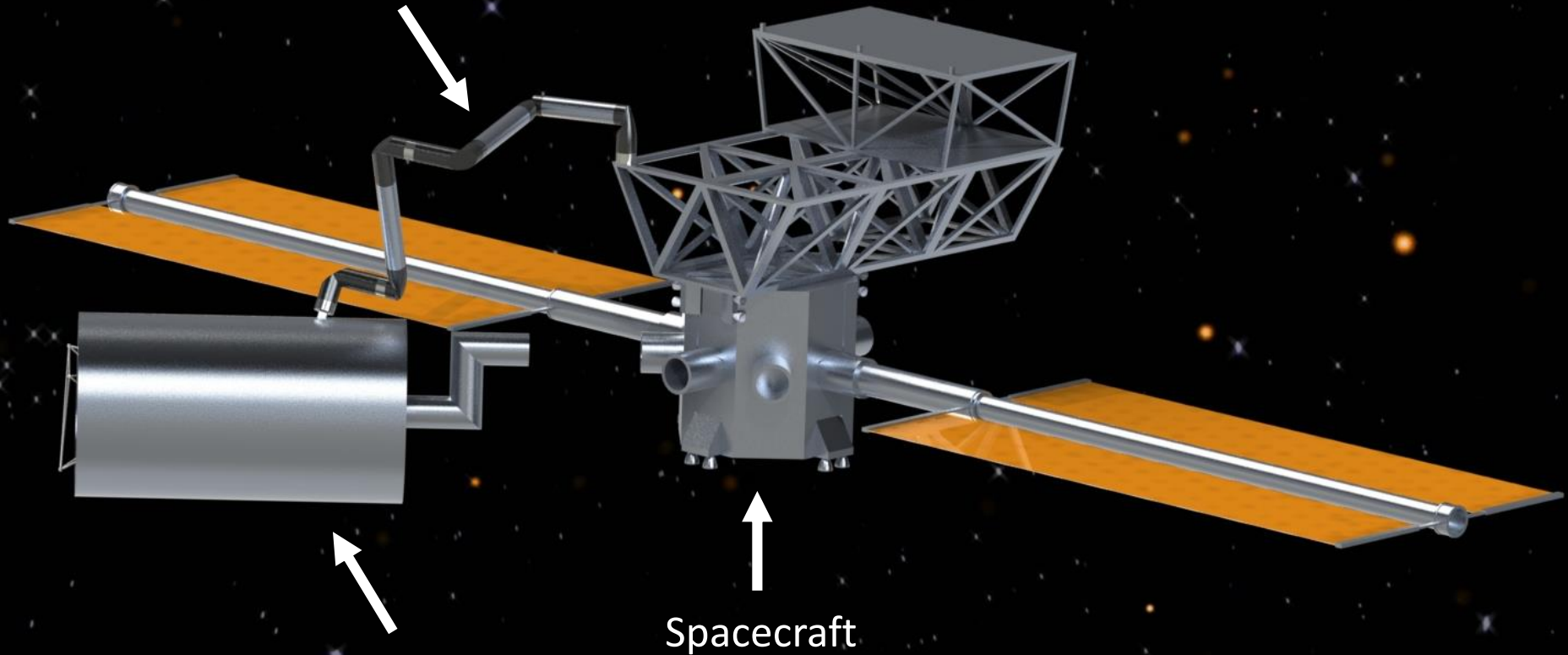
Telescope's spacecraft bus as the assembly platform

Assembly Platform



# Key Aspects of the iSAT Paradigm

(4) Supervised autonomous robotic arm



- (1) Cargo delivery vehicle
- (2) Multiple launch vehicles
- (3) Rendezvous and proximity operations

# Existing Cargo Delivery Vehicles

Rendezvous, Grappled, and Berthed



Northrop Grumman's Cygnus

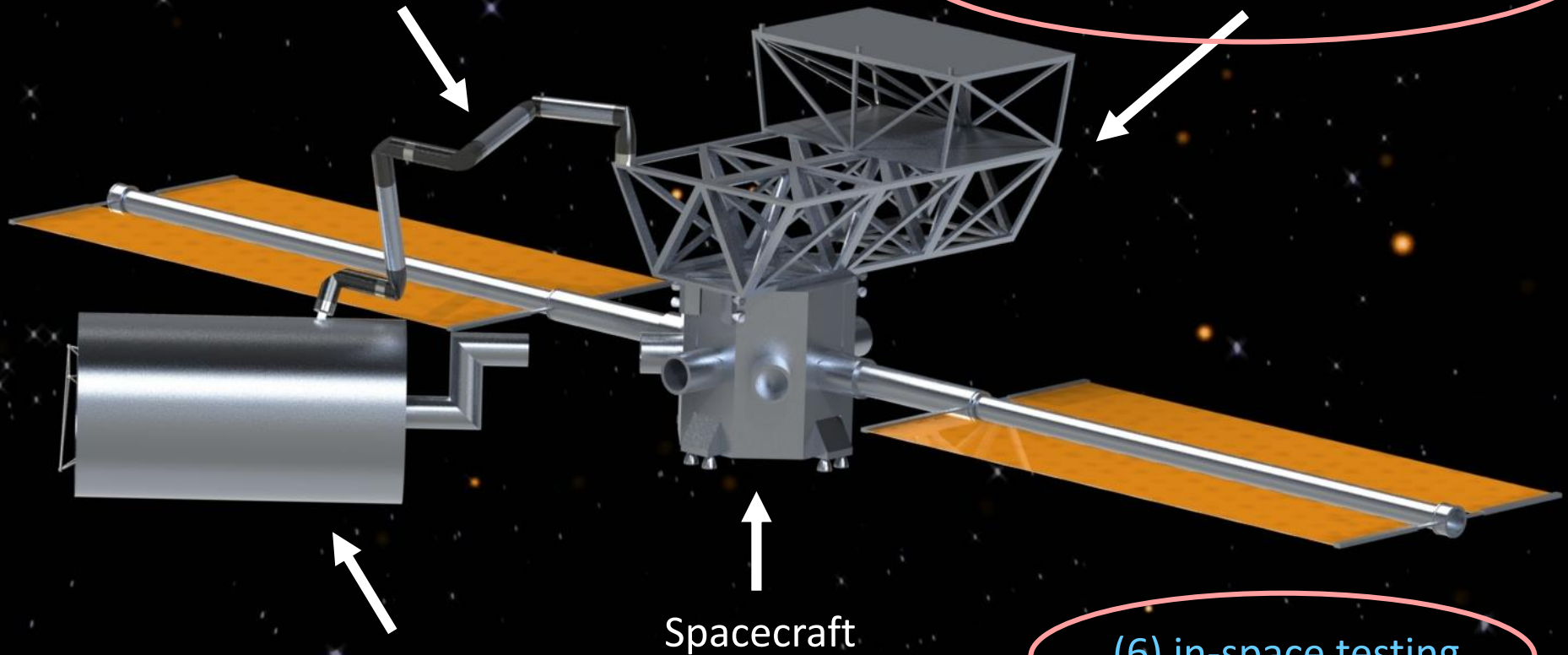


SpaceX's Dragon

# Key Aspects of the iSAT Paradigm

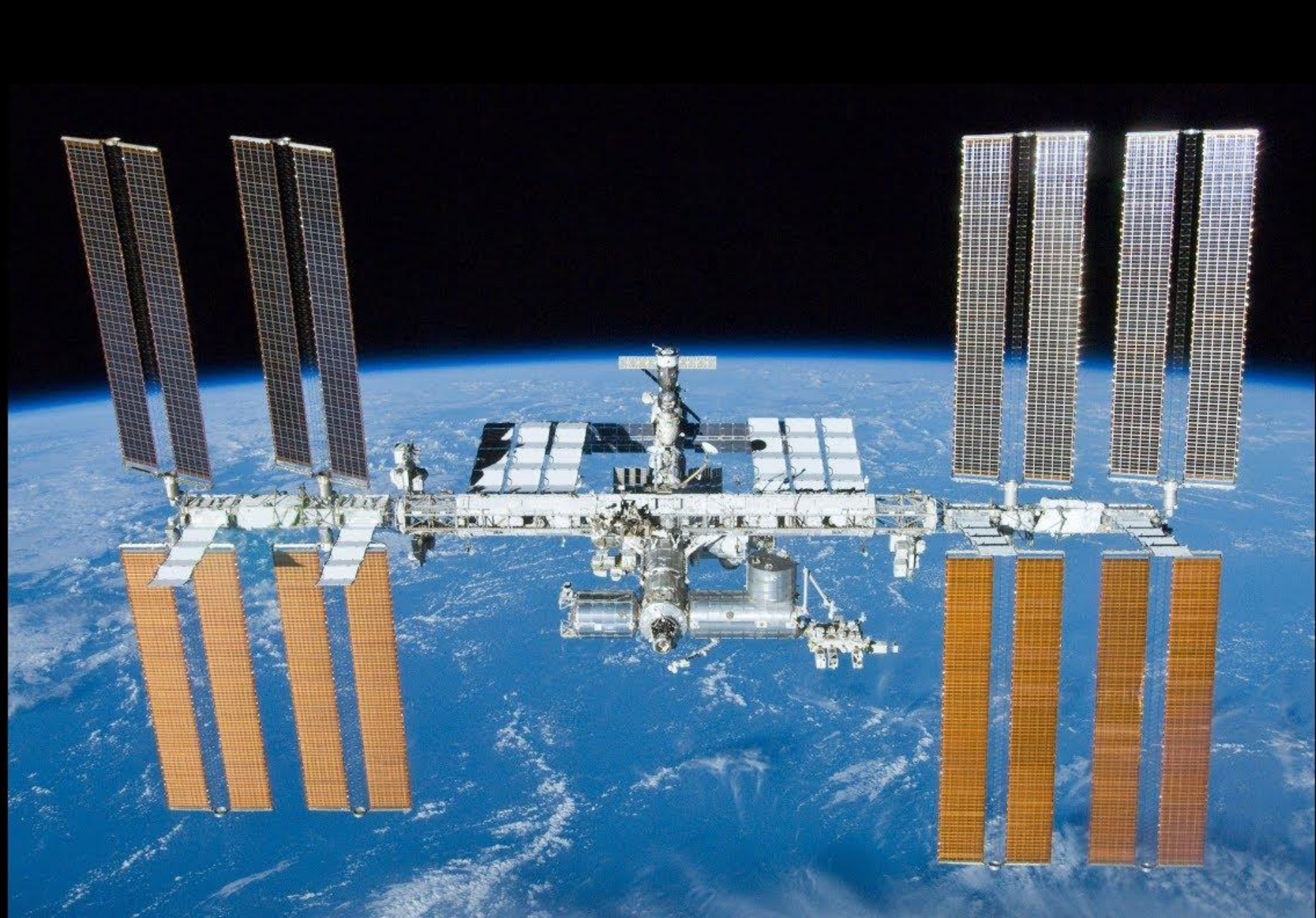
(4) Supervised autonomous robotic arm

(5) Modularized flight elements



- (1) Cargo delivery vehicle
- (2) Multiple launch vehicles
- (3) Rendezvous and proximity operations

(6) in-space testing



# Delivery ConOps

## Disposable Cargo Delivery Vehicle (CDV)

Observatory spacecraft bus and robotics on orbit

CDV RPO Grappled by Assemblage

Assemblage robotics berth CDV, remove cargo, releases CDV

Observatory Maneuver to SEL2

CDV maneuver to acquire assembly orbit

Empty CDV Disposal to Helio-centric

CDV Separation

Fairing Separation

2<sup>nd</sup> Stage Disposal to heliocentric

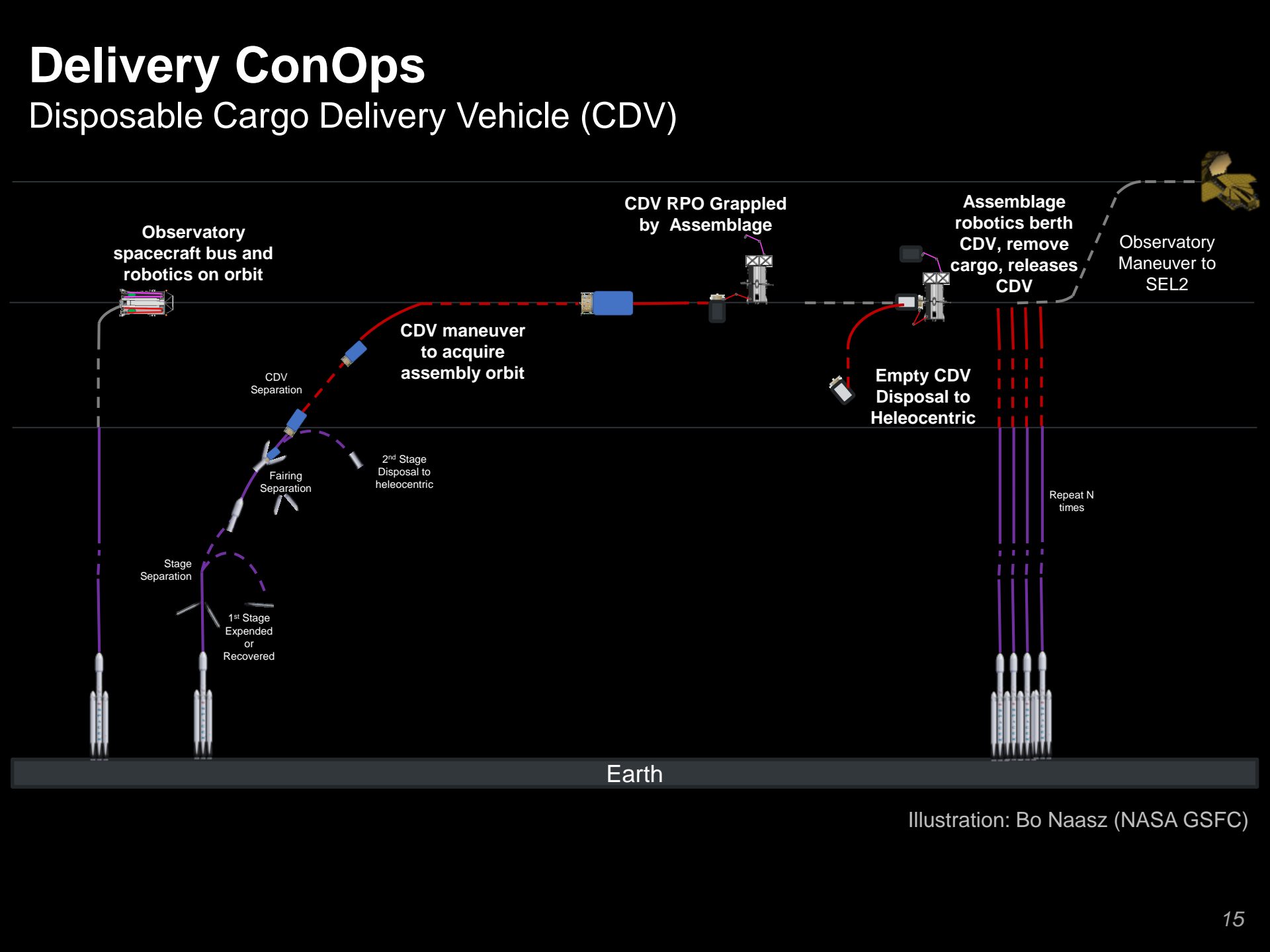
Stage Separation

1<sup>st</sup> Stage Expended or Recovered

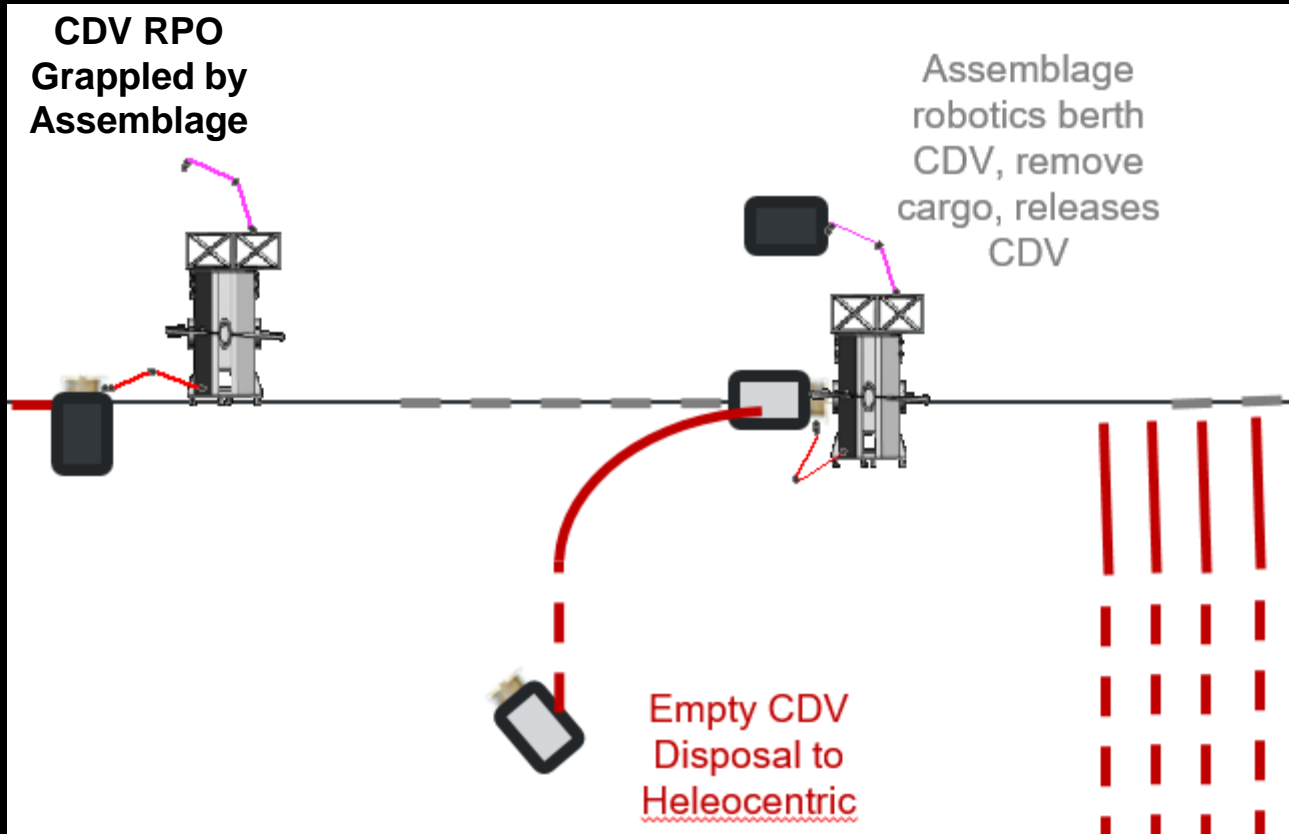
Repeat N times

Earth

Illustration: Bo Naasz (NASA GSFC)



# Delivery Via Disposable Cargo Delivery Vehicle





# Delivery ConOps

## Disposable Cargo Delivery Vehicle (CDV)

Observatory spacecraft bus and robotics on orbit

CDV RPO Grappled by Assemblage

Assemblage robotics berth CDV, remove cargo, releases CDV

Observatory Maneuver to SEL2

CDV maneuver to acquire assembly orbit

Empty CDV Disposal to Helio-centric

CDV Separation

Fairing Separation

2<sup>nd</sup> Stage Disposal to heliocentric

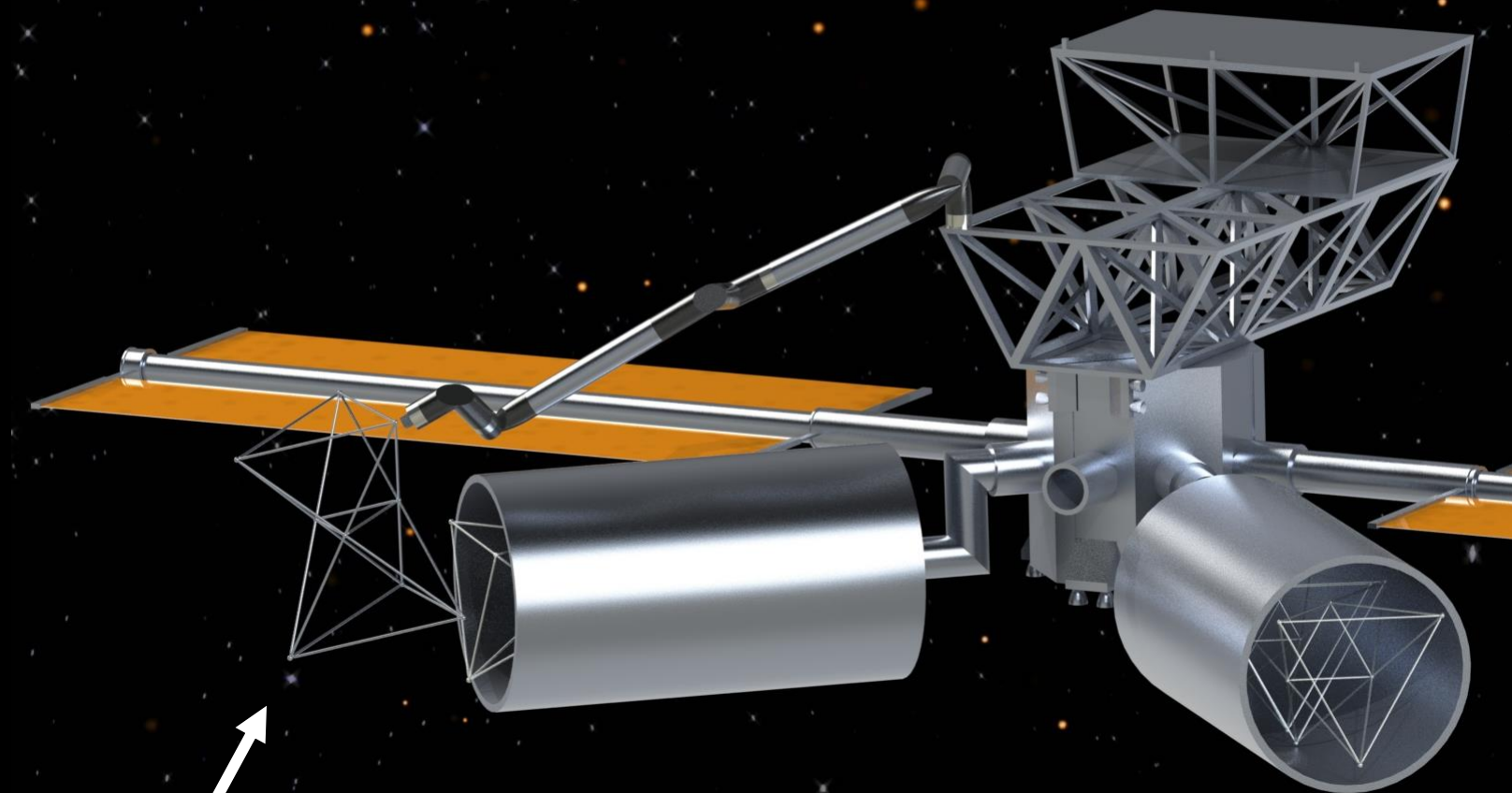
Stage Separation

1<sup>st</sup> Stage Expended or Recovered

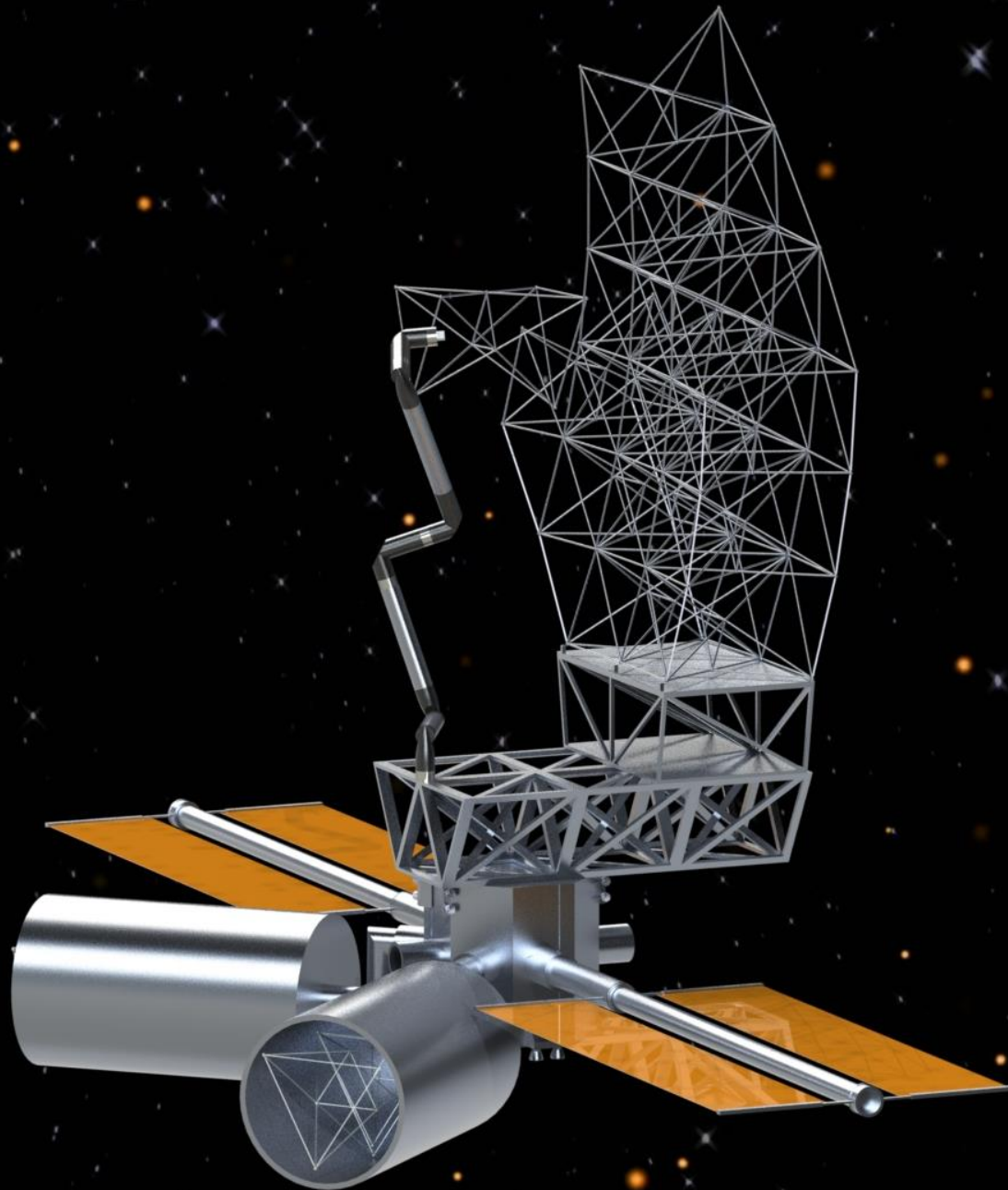
Repeat N times

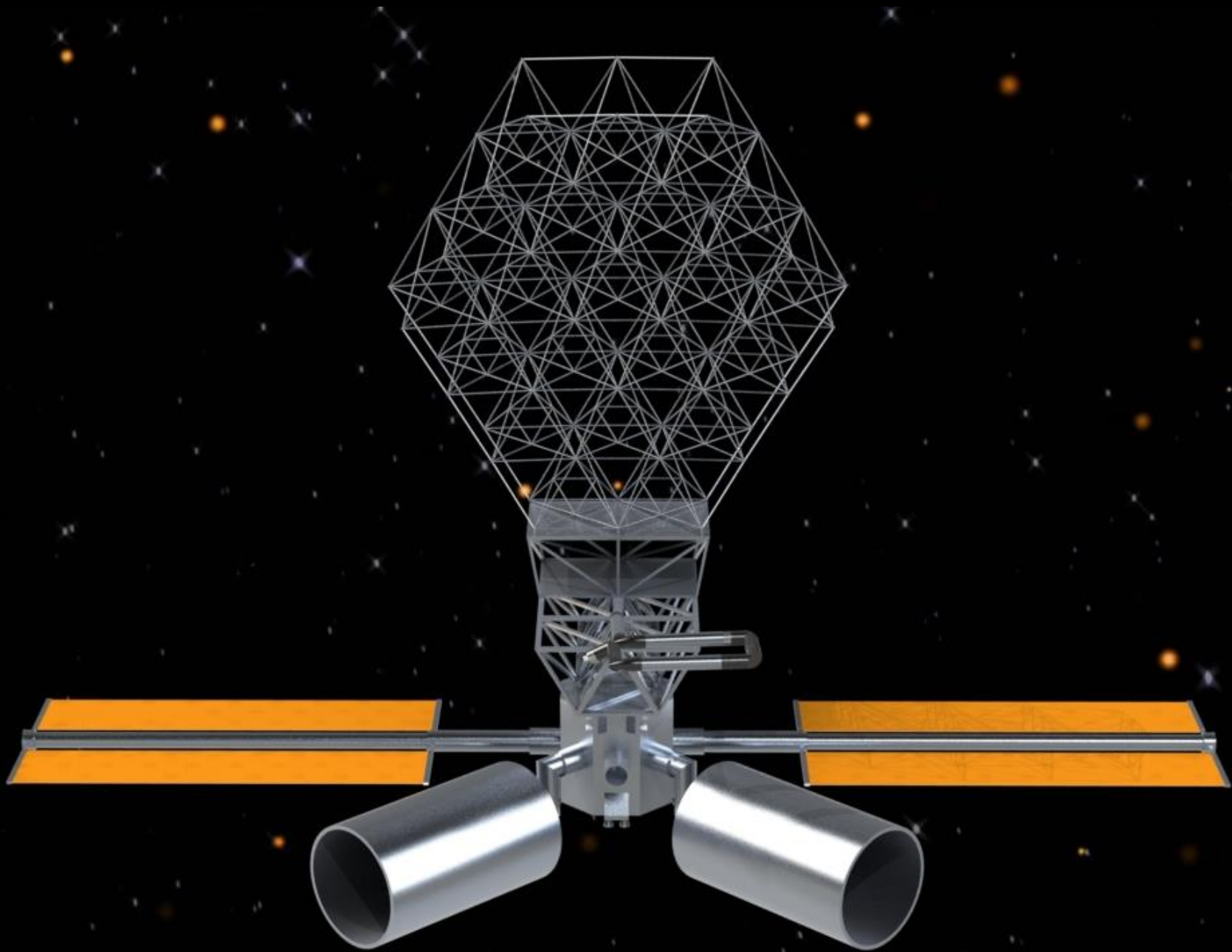
Telescope PM Size (m)	Total Launches
5	2
10	4
15	7
20	12

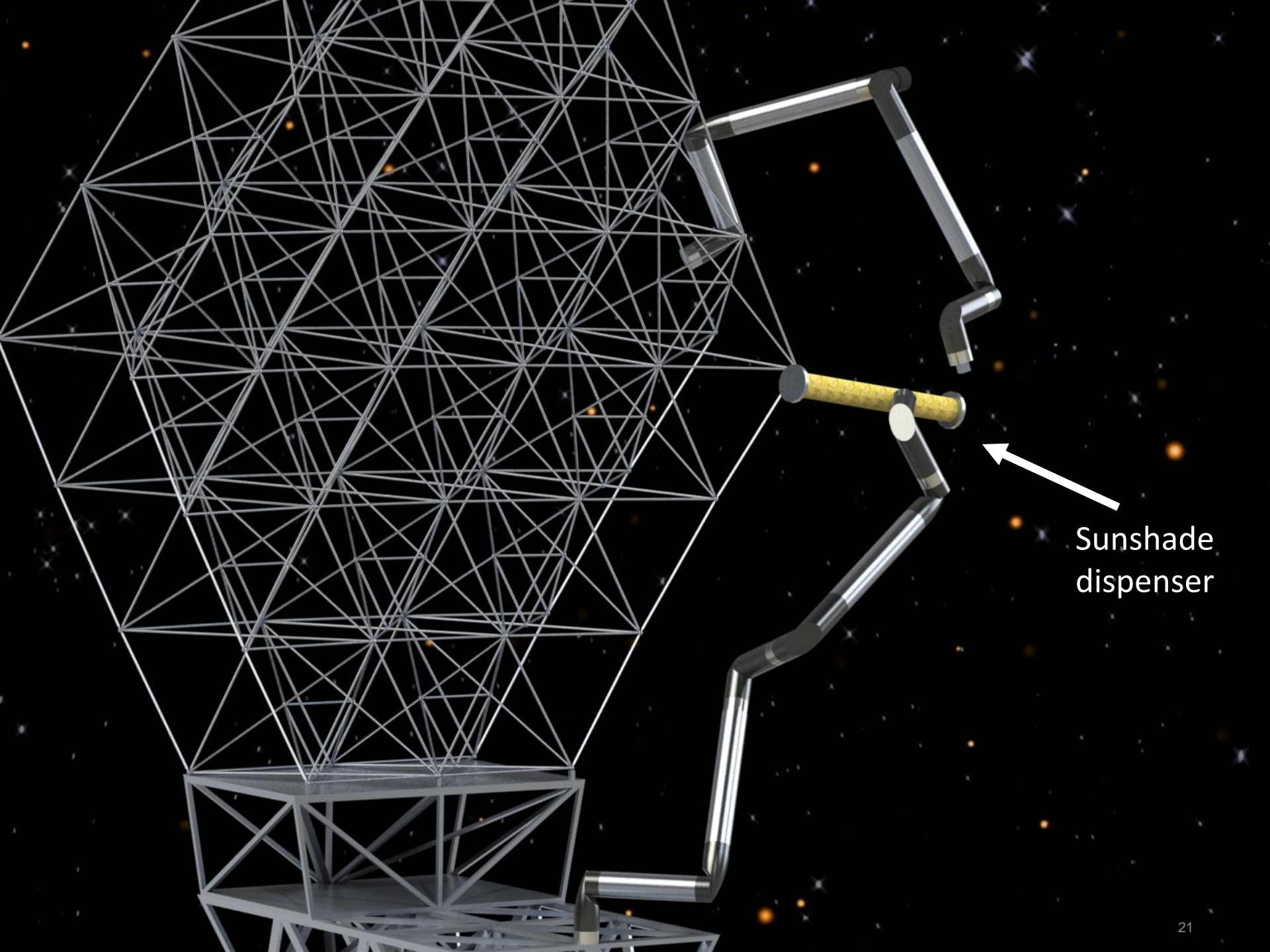
Illustration: Bo Naasz (NASA GSFC)



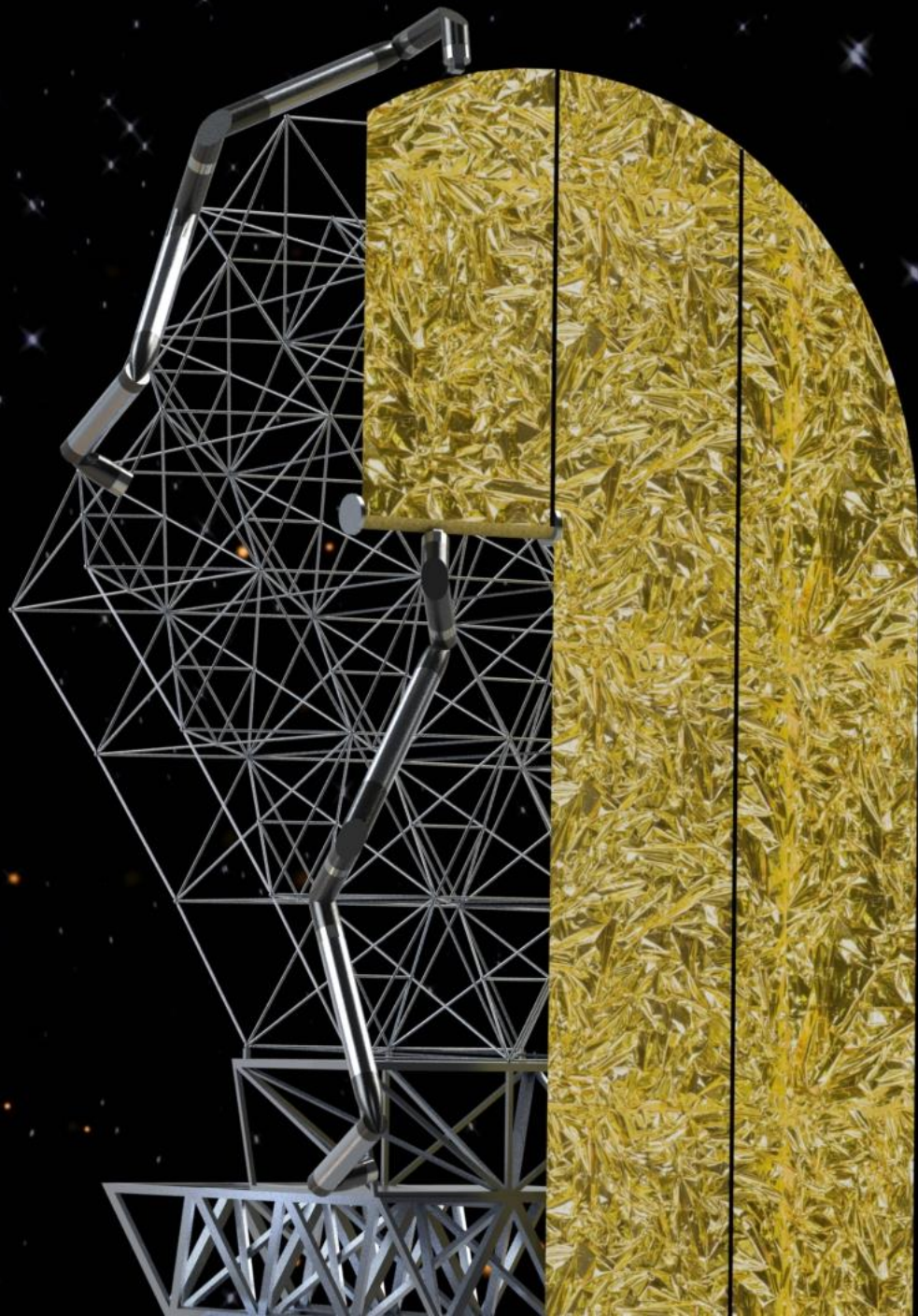
telescope  
backplane truss

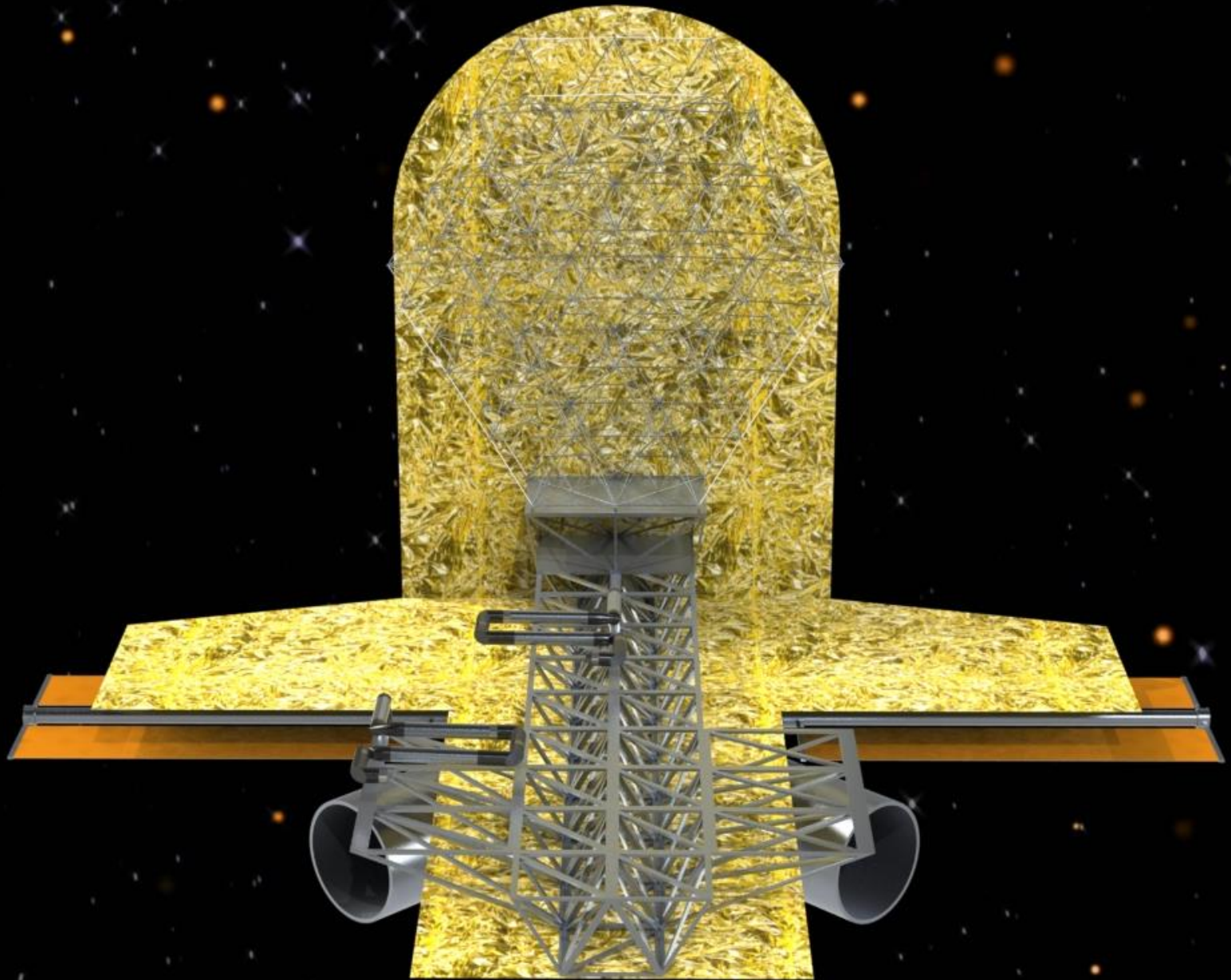




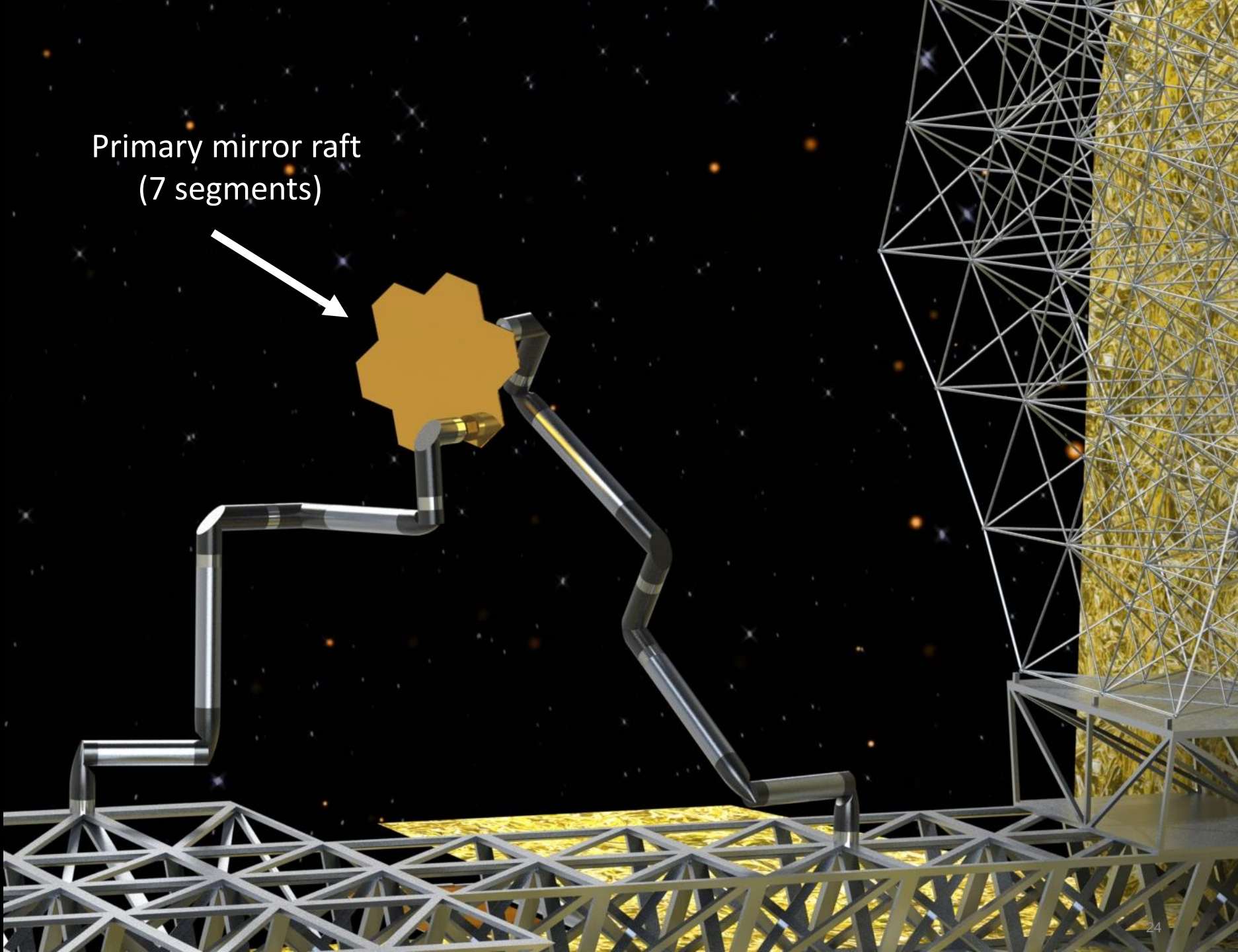
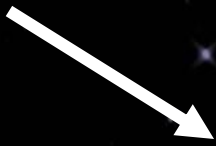


Sunshade  
dispenser

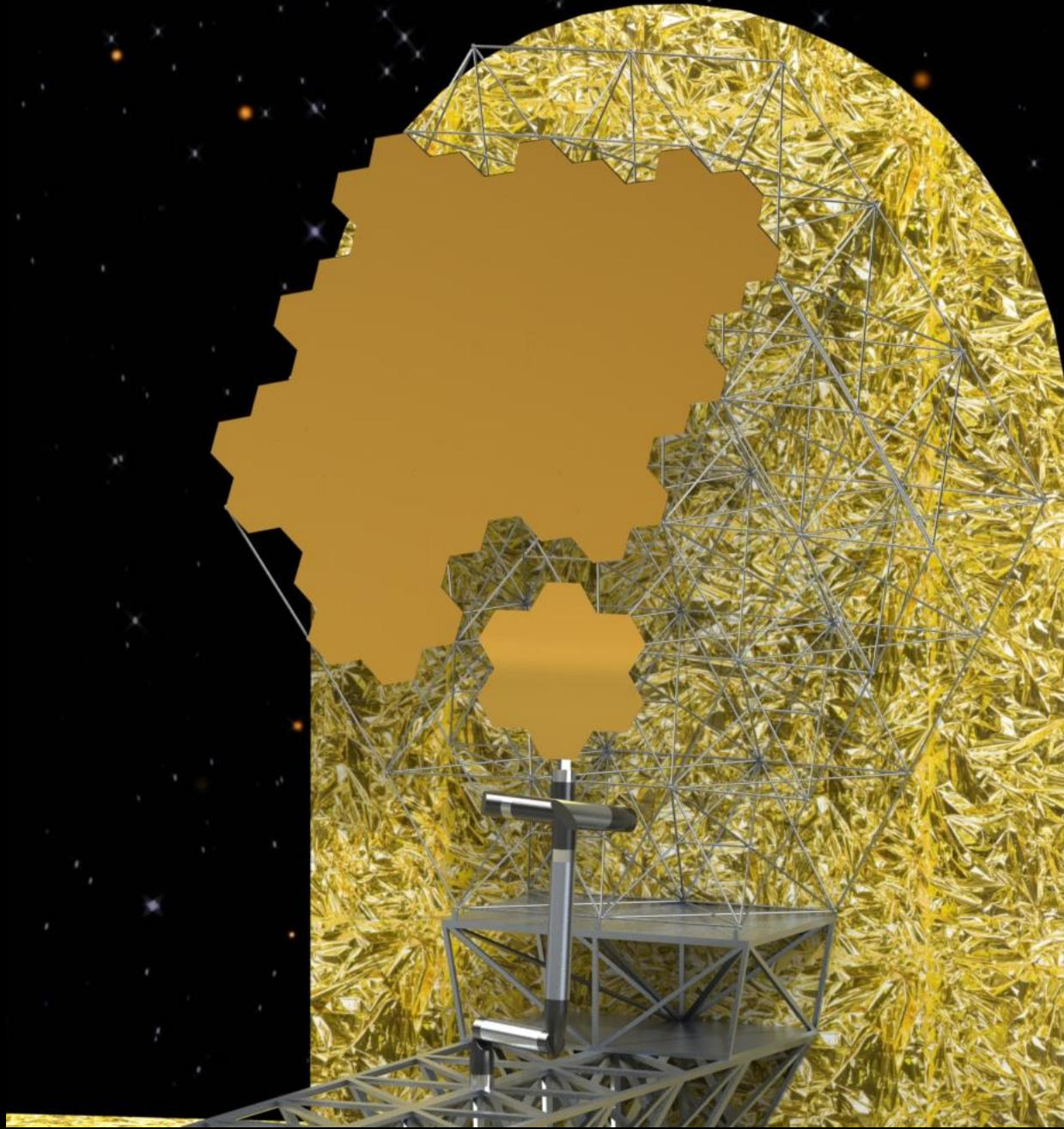




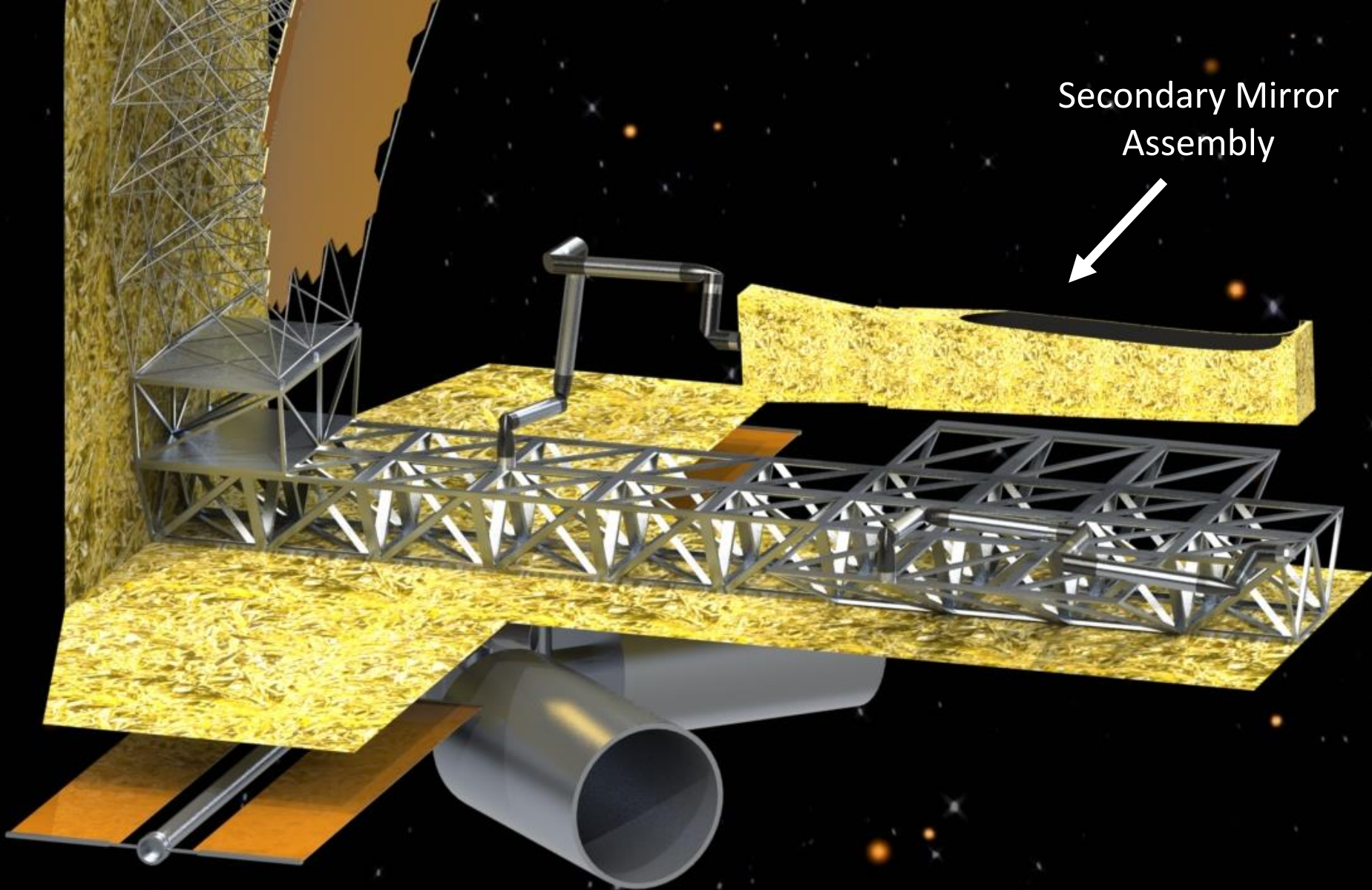
Primary mirror raft  
(7 segments)



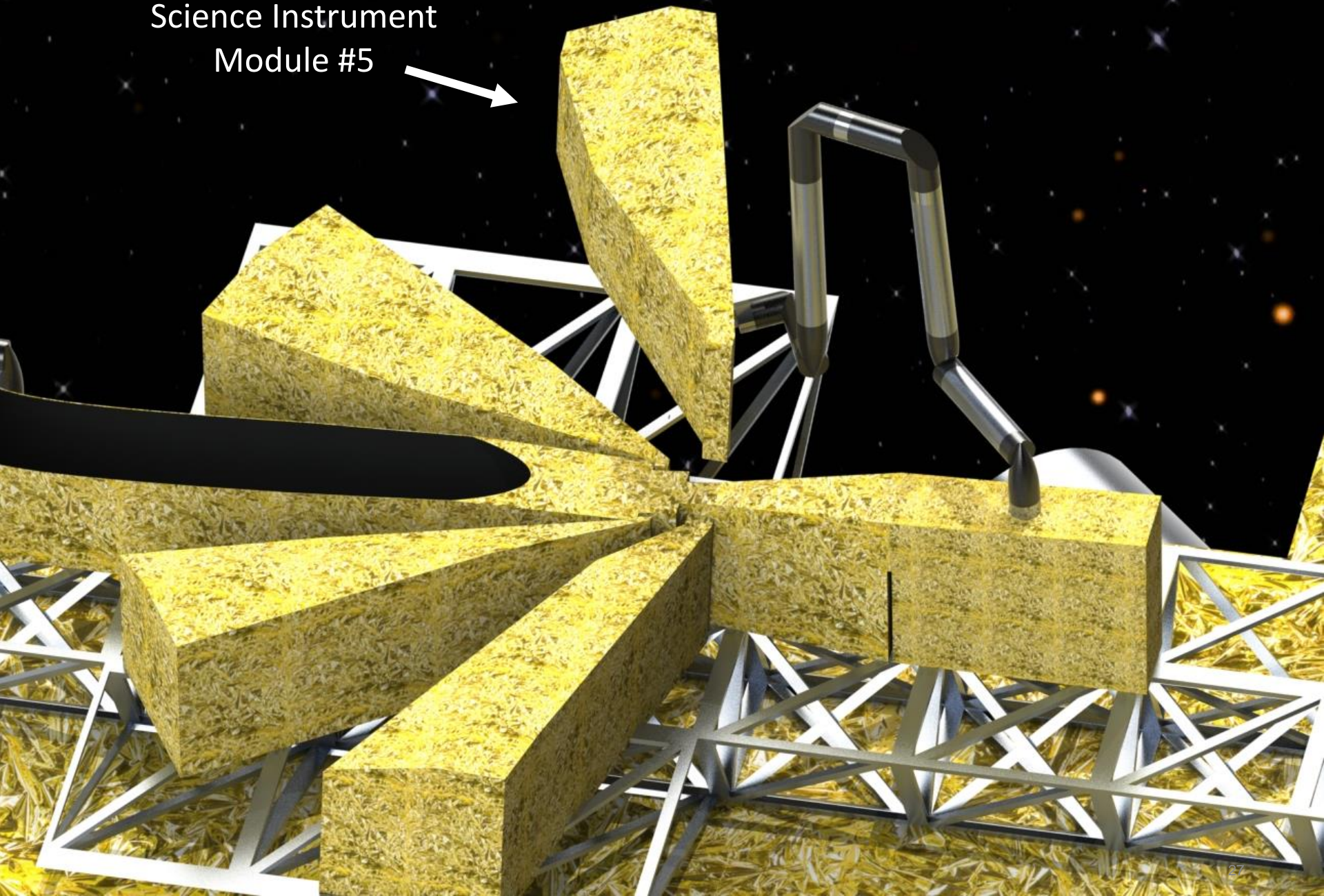




Secondary Mirror  
Assembly



Science Instrument  
Module #5



# iSAT Technical Challenges

(incomplete list)

Stiff linear structures and precision joining mechanisms

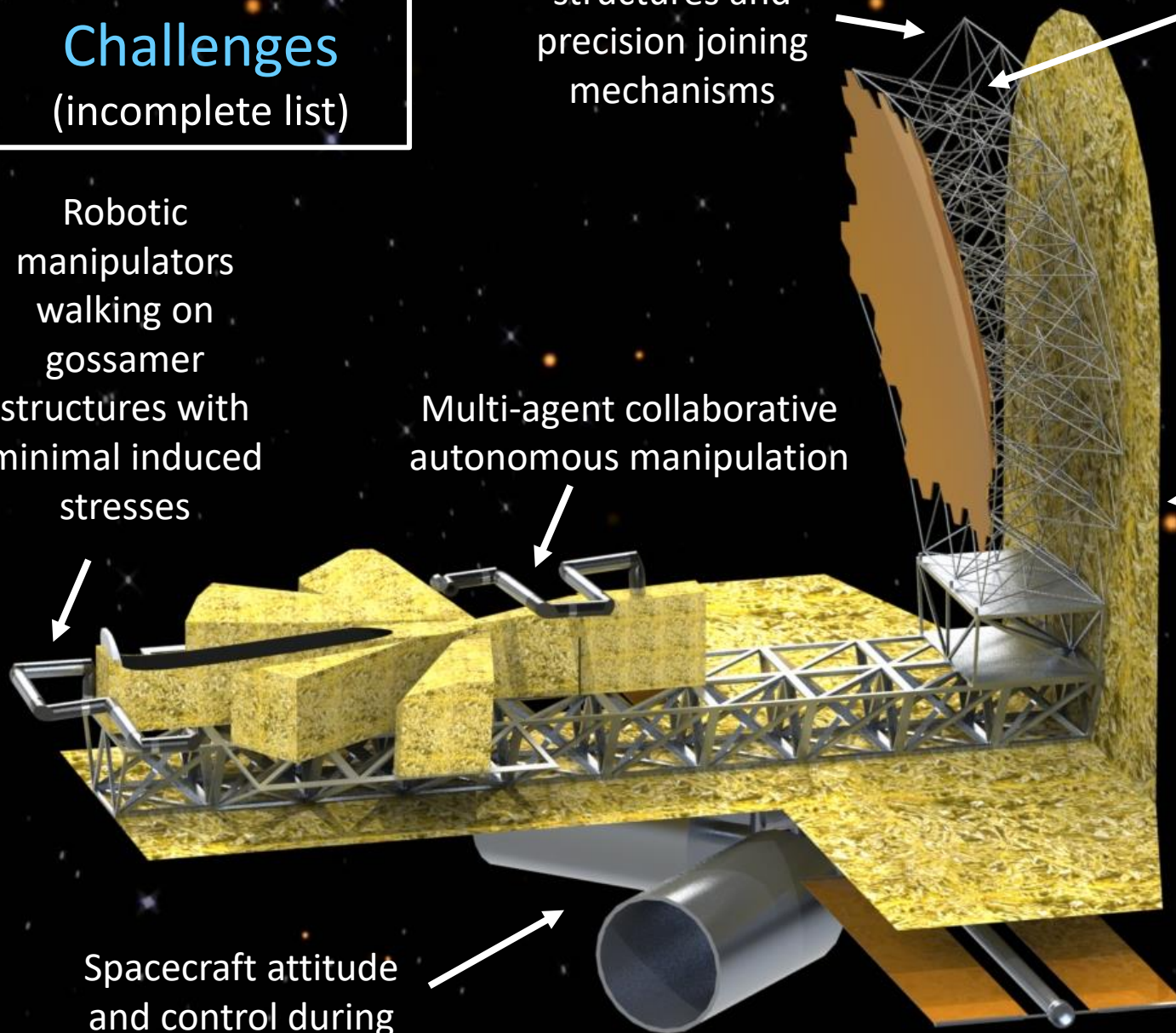
Precise, adjustable, reversible interfaces with harnesses

Robotic manipulators walking on gossamer structures with minimal induced stresses

Multi-agent collaborative autonomous manipulation

Assembly and manipulation of soft goods such as Mylar sheets and blankets

Spacecraft attitude and control during assembly



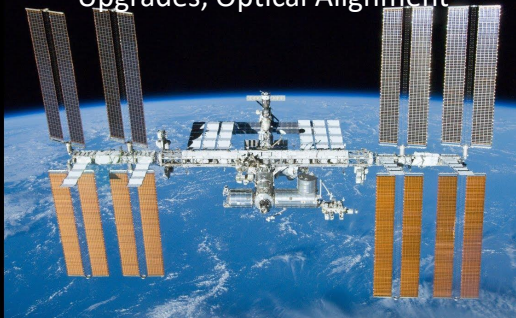
# Study Key Findings

# iSAT Leverages Many TRL 9 Capabilities

## Past Capability Advances



**HST Servicing** – Inspects, Repairs, Upgrades, Optical Alignment



**ISS Assembly** – Modularity, Multiple LV's, Robotic Arms



**Orbital Express**

Autonomous Rendezvous and Soft Capture, Removal/installation of ORUs, Fluid Transfer

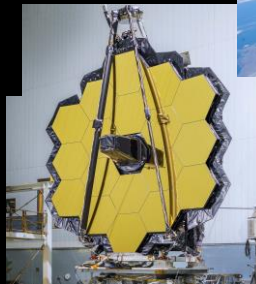
## Ongoing Capability Improvements



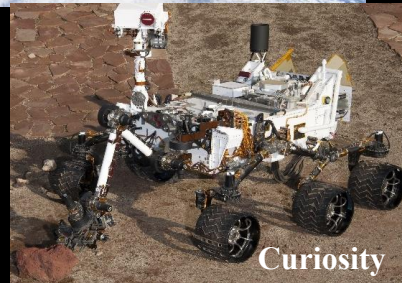
**ISS Servicing and Assembly** – Robotic Repairs, Autonomous Docking, Instrument Assembly



**Space X Dragon Resupply**



**JWST:**  
Segmented Optics  
WFS&C Phasing

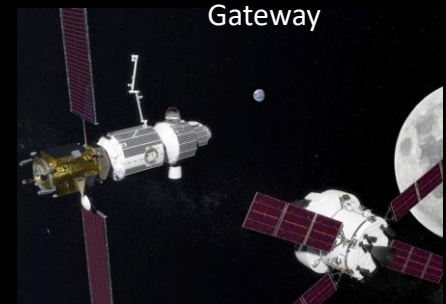


**Curiosity**  
**Supervised Autonomy Robotics**

**Commercial LEO** – Infrastructure Buildup, Support Services

## Future Capability

**Advanced Servicing** –  
Autonomy, Telerobotics,  
Refueling, Servicing



**Gateway**



**Restore-L**

**Mars Sample Return**



# No technical showstoppers

Further engineering development required; several technology gaps

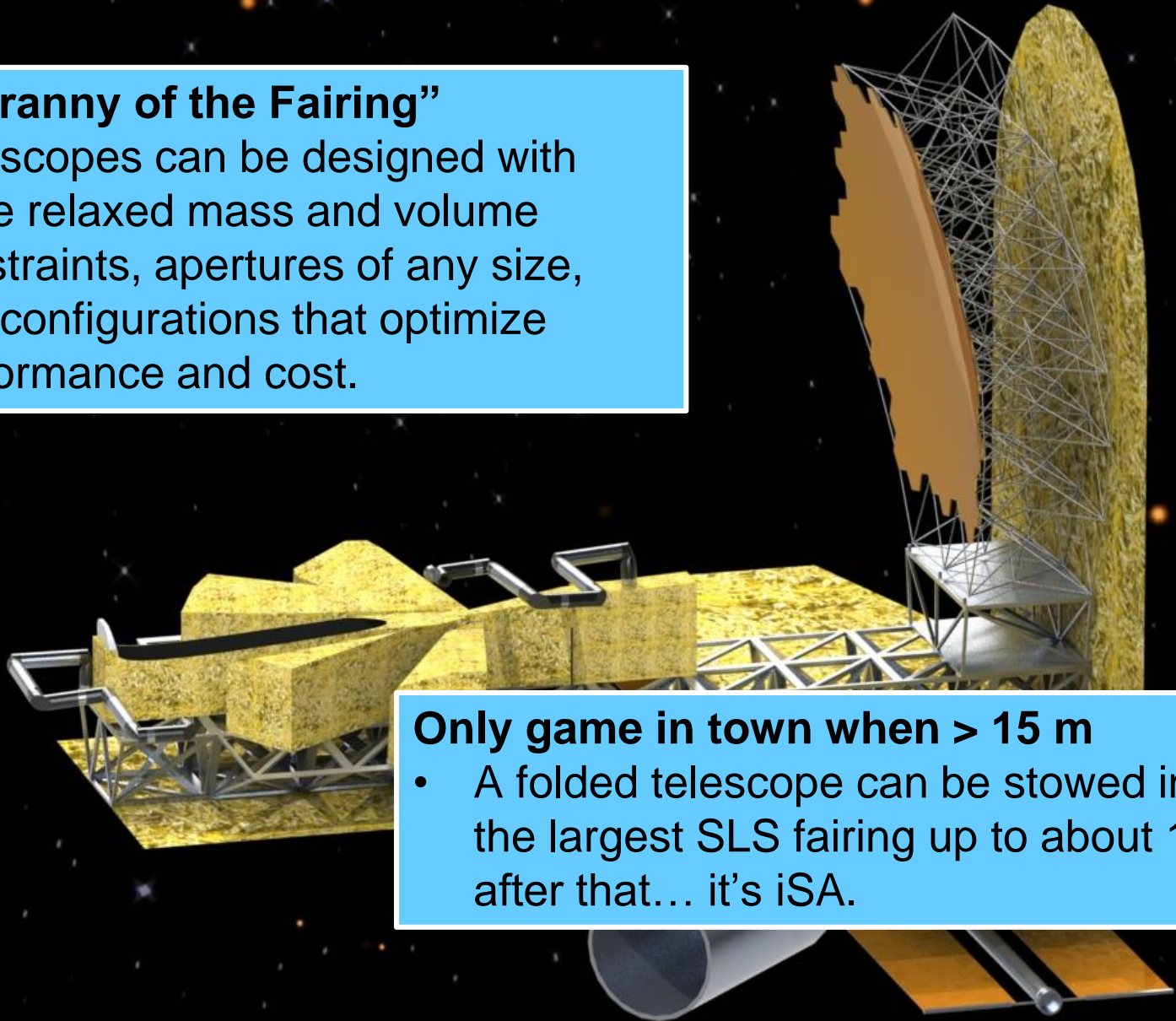
#	ISA Key Capabilities	Status	Representative Examples	Readiness for Observatory ISA
1	<b>Modular Elements</b>	Flight Demonstrated	Instruments on HST, instruments installed on ISS	<b>Low</b>
		Active Development	JWST primary mirror segments	
2	<b>Launch Vehicles</b>	Flight Demonstrated	SpaceX Falcon, Falcon Heavy, ULA's Delta IV	<b>High</b>
		Active Development	SLS, Blue Origin, Starship, Vulcan Centaur	
3	<b>RPO</b>	Flight Demonstrated	DARPA Orbital Express, NASA OSIRIS-Rex, Cygnus, Dragon, Crew Dragon, ATV, HTV, Progress, Soyuz	<b>High</b>
4	<b>CDVs</b>	Flight Demonstrated	SpaceX Dragon, Cygnus from Northrop Grumman	<b>High</b>
5a	<b>Space Robotics Hardware</b>	Flight Demonstrated	Several robotic arms on ISS (e.g. Canadarm 2), Orbital Express robotic arm, Mars Rover arms, Shuttle arm	<b>High</b>
		Active Development	NASA Restore-L and DARPA RSGS robotic servicing arms, Canadarm 3, Maxar's Dragonfly arm, Mars 2020 rover	
5b	<b>Space Robotics Software</b>	Flight Demonstrated	Mars Rover Autonomy (e.g. MSL, MER), ISS, Orbital Express	<b>Medium</b>
		Active Development	Mars 2020, Mars Sample Return, NASA Restore-L, DARPA RSGS, NASA Tipping Point Demonstrations	
6	<b>In-space Verification and Validation</b>	Flight Demonstrated	Instruments on HST, instruments installed on ISS	<b>Low</b>
		Active Development	JWST primary mirror segments and wavefront control	

Technology investments needed most in items 1 and 6

# Mass and Volume Constraints Decoupled

## No “Tyranny of the Fairing”

- Telescopes can be designed with more relaxed mass and volume constraints, apertures of any size, and configurations that optimize performance and cost.



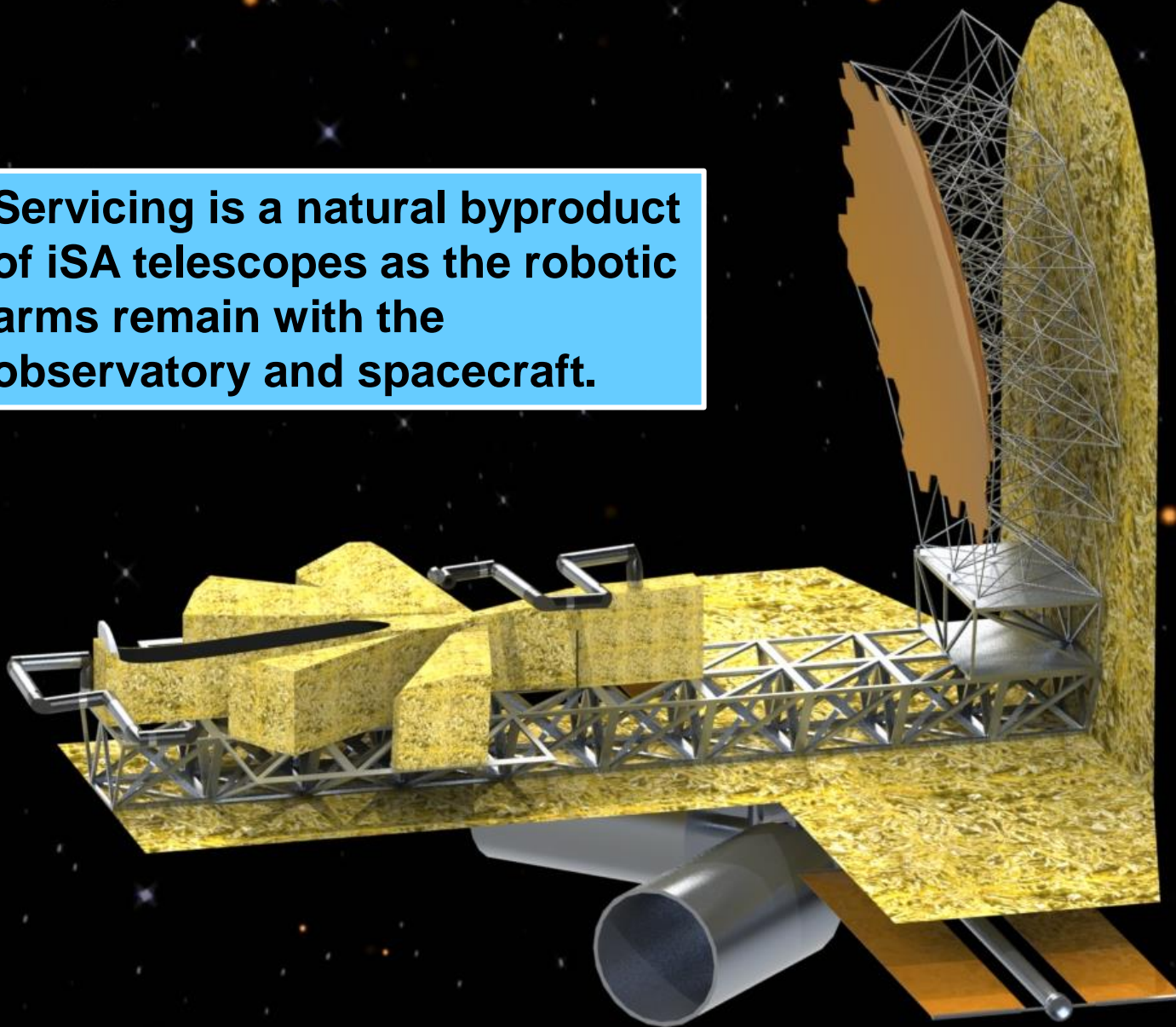
## Only game in town when $> 15$ m

- A folded telescope can be stowed into the largest SLS fairing up to about 15 m, after that... it's iSA.



# iSA Solves the in-Space Servicing Question

**Servicing is a natural byproduct of iSA telescopes as the robotic arms remain with the observatory and spacecraft.**



# Welcomed, but not required...

- **Future space infrastructures (e.g. Lunar Gateway, ISS)**
- **Astronauts**
- **Large future launch vehicles (e.g. BFR, New Glenn, SLS)**



# Key **Science** Benefits Enabled by iSA

- No “Tyranny of the fairing”
  - Telescope diameters and configurations that achieve science goals not possible with apertures constrained by single launches
  - Instruments may be more capable as they are independently launched and less constrained by mass and volume
- Telescopes can evolve and last decades
  - Continuous stream of planned instrument upgrades (e.g., HST)
  - Can plan for refueling and preventive maintenance missions that extend useable lifetime
  - Can authorize unexpected repair missions
- No explicit servicer needed
  - Cost and science benefits



# iSA can Reduce Mission Risk

- Eliminates complex autonomous self-deployments
- Mitigates the risks associated with a deployment anomaly
  - ❖ Faulty modules can be replaced during commissioning
  - ❖ Or during operations, with servicing (second chances)
- Mitigates the risks associated with a single launch vehicle
  - ❖ *Launch failure need not be mission failure*



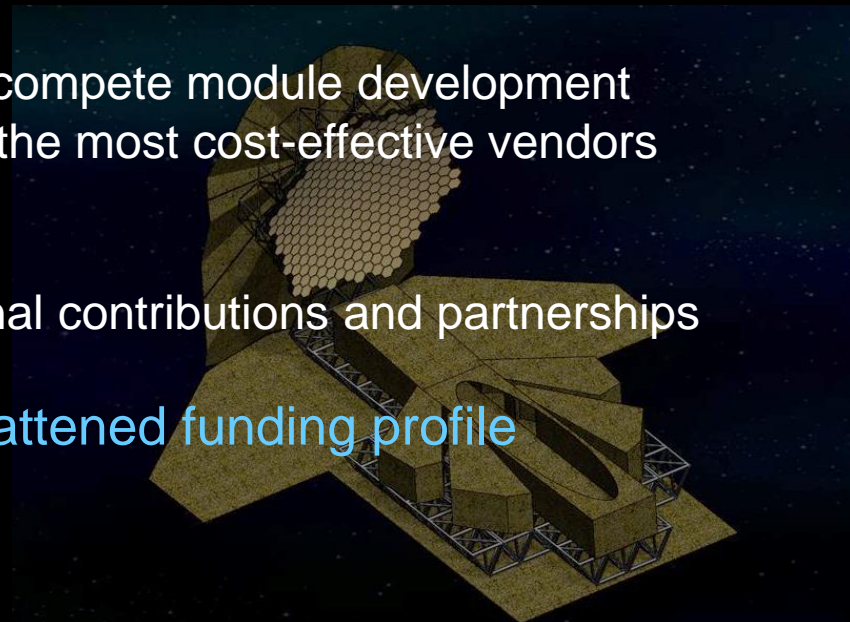
# Key **Cost** Benefits Enabled by iSA (1 of 2)

- **Relaxes mass and volume constraints**
  - Reduces engineering design complexity and time (i.e. cost)
  - Eliminates complex folding designs, reduces mass iterations, less need for complex modeling
- **More versatile scheduling**
  - More work conducted in parallel
  - Multiple parallel deliveries (swim lanes) so AIT team can move to different module deliveries when there are schedule delays (and not turn into a large marching army)
- **Modules with standardized interfaces help speed up AIT, especially during anomaly resolution**
- **Eliminates costly systems-level testing activities**
  - Enabled by greater degrees of designed on-orbit adjustability and correctability to meet system tolerance requirements



# Key **Cost** Benefits Enabled by iSA (2 of 2)

- Diminishes cost and schedule impacts from late-stage hardware re-design changes and iterations.
- Reduces need for ruggedizing the system and its interfaces to survive launch
- Less need for new and larger ground test facilities
- **Spread the wealth:** Can distribute and compete module development work across NASA and industrial base to the most cost-effective vendors and facilities
- **Share the wealth:** Enhances international contributions and partnerships
- More readily enables prescribed or flattened funding profile programs



# Cost Estimation

ISA will incur additional cost compared to a conventional, single launch observatory. These include:

- Modularity, multiple launches, cargo delivery vehicles, rendezvous and proximity operations, assembly robotics

ISA will likely offer opportunities for cost savings in the development of flight system elements such as the telescope, instruments, spacecraft

- These elements typically represent 60-70% of mission costs. Hence, this can be a source of significant savings.
- Flight system assembly, I&T are other areas of potential savings.

→ What is the net effect?

WBS 1-3 Mng. SE. SMA	WBS 4 SCI	WBS 5.1 Telescope Structure	WBS 5.2 Telescope Optics	WBS 5.3 Sunshade	WBS 5.4 Inst	WBS 5.5 Robotics	WBS 6 SC	WBS 7-9 MOS/GDS	LV	CDV	Ops	WBS 10 SI&T

Caption: Relative cost comparison between single-launch vehicle observatory and iSAT. Green represents WBS elements where ISA may provide cost benefits while red represents elements where ISA may have a cost increase in comparison to a conventional, single-launch approach

# Study Conclusions

- Even telescopes with apertures as small as 5 m may benefit from iSA as the implementation approach due to **potentially better risk postures** and opportunities for **potential cost savings** in comparison to the conventional approach of deployment from a single launch.
- When including future **servicing missions** the benefits may be even more important.

***Actual cost and risk differences will depend ultimately on the mission and point design selected.***



# Suggestions to NASA

If the Astro2020 Decadal Survey recommends a large space observatory, we suggest:

1) *NASA (a) conduct a **detailed study** of an iSA implementation of the specific observatory and (b) **trade** it against the conventional single launch approach*

2) *and*

*NASA **initiate a technology development program** to reduce the technology gaps associated with in-space assembled observatories*

# Moving Forward

- iSA has made significant progress over the last 15 years to the point it can now be considered as an alternative implementation approach to realize large space telescopes.
  - ❖ *“another tool in the tool box”*
- Designers can now consider hybrid solutions to reduce cost and risk (some elements deployed, some elements assembled).
- Next step is to focus on technology gaps and technologies:
  - ❖ Develop technology roadmaps
  - ❖ Recommend risk reduction demonstrations for future flight mission

**Please contact the Study leads if you are interested in advancing this work.**

# More Information on our Website

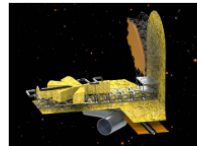
[https://exoplanets.nasa.gov/exep/technology/in-space-assembly/iSAT\\_study/](https://exoplanets.nasa.gov/exep/technology/in-space-assembly/iSAT_study/)

## NASA in-Space Assembled Telescope (iSAT) Study

NASA in-Space Assembled Telescope (iSAT) Study | Steering Committee Telecons  
| Study Workshops and F2F Meetings | Study Member Telecons

Large aperture telescopes benefit all astrophysics as well as planetary and Earth science. They provide unprecedented spatial resolution, spectral coverage, and signal to noise, advancing all of these science areas. Envisioning the need for future large segmented telescopes to one day exceed the fairing size of existing or even planned launch vehicles. NASA will need to begin considering the in-space assembly (ISA) of these future assets. In addition, robotically assembling space telescopes in space rather than deploying them from single launch vehicles offers the possibility, in some circumstances, of reduced cost and risk for even smaller telescopes. This possibility, however, has not been proven. Therefore, following discussions within NASA's Science Mission Directorate (SMD) and Astrophysics Division (APD), the SMD Chief Technologist and APD Division Director have commissioned a study to assess the cost and risk benefits, if any, of the ISA of space telescopes.

In particular, the study must answer the question: *"When is it advantageous to assemble space telescopes in space rather than to build them on the Earth and deploy them autonomously from individual launch vehicles?"* The Study Charter (on the right) describes the plan for the study deliverables, process, and membership. The goal for completion of the study is May 2019 culminating in a submitted whitepaper to the National Academies' 2020 Astronomy & Astrophysics Decadal Survey.



iSAT ConOps Graphical Storyboard: 20m Segmented UV/V/NIR telescope

Study Area	Key Findings	Recommendations
Study Area 1	Key findings regarding the benefits of in-space assembly for large aperture telescopes.	Recommendations for future studies and development.
Study Area 2	Key findings regarding the challenges of in-space assembly.	Recommendations for risk mitigation and cost reduction.
Study Area 3	Key findings regarding the technical feasibility of in-space assembly.	Recommendations for technology development and testing.
Study Area 4	Key findings regarding the operational requirements for in-space assembly.	Recommendations for mission design and operations.
Study Area 5	Key findings regarding the economic viability of in-space assembly.	Recommendations for funding and resource allocation.

iSAT: Benefits and Challenges Table

### iSAT Astro2020 Whitepaper

When is it Worth Assembling Observatories in Space?  
Astro2020 APC Whitepaper  
Study of In-Space Assembly Options  
Prepared for the Astrophysics Division (APD) by the Study of In-Space Assembly (ISA) Study

Contributors: [List of names]



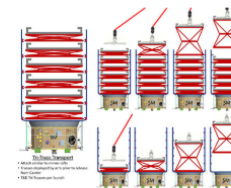
When is it Worth Assembling Observatories in Space? Astro 2020 APC Whitepaper, Mukherjee et al. 2019

### iSAT Study Final Report



View Report PDF

### iSAT Study Additional Information



View PDF